# Moving Mesh Methods with Upwinding Schemes for Time-Dependent PDEs

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It is well known that moving mesh and upwinding schemes are two kinds of techniques for tracking the shock or steep wave front in the solution of PDEs. It is expected that their combination should produce more robust methods. Several upwinding schemes are considered for non-uniform meshes. A self-adaptive moving mesh method is also described. Numerical examples are given to illustrate that in some cases, especially for hyperbolic conservative laws with nonconvex flux, the upwinding schemes improve the results of the moving mesh methods. Comparing the results of several upwinding schemes, we find the local piecewise hyperbolic method (PHM) is very efficient and accurate when combined with a moving mesh strategy. © 1997 Academic Press

## **1. INTRODUCTION**

Moving mesh methods are becoming increasingly popular for several kinds of parabolic and hyperbolic partial differential equations (PDEs) involving fine scale structures such as steep moving fronts, emerging steep layers, pulses, shocks, etc. Moving mesh methods use nonuniform spatial grids and move the grid continuously in the spacetime domain. The discretization of the PDE and the grid selection procedure are intrinsically coupled.

In this paper we investigate moving mesh methods applied to PDEs of the form

$$\frac{\partial u}{\partial t} = \frac{\partial f(u)}{\partial x} + \mu u_{xx} + h(u), \qquad (1)$$

where f(u), h(u) are differentiable functions. This equa-

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tion, which is often called a reaction–convection–diffusion equation, is representative of many problems which are solved by moving mesh methods in one dimension.

Much work has been devoted to the construction of shock capturing schemes on uniform grids, such as Godunov, MUSCL, PPM, FCT, ENO, and PHM (see Salari and Steinber [21] and its references, Shu and Osher [22], and Marquina [16] for more information). Most of these schemes possess the total variation diminishing (TVD) feature which is necessary for high order oscillation-free schemes. The conventional scheme on a uniform grid needs enough points to capture the shock or steep wave front, which often puts too many points in the area where the solution is very flat and smooth. Moving mesh methods, on the other hand, can redistribute the nodes to those areas where the solution varies rapidly at each time step to gain computational efficiency. Thus, many useful schemes have been applied on moving grids in recent years. Hyman and Harten [8] applied the Godunov method on a nonuniform grid where the grids move along the characteristic direction. Salari and Steinberg [21] developed a FCT method on a moving grid (FCTMG) based on adaptive grid generation algorithms. Zarnowski [26] introduced a moving grid method for scalar conservation laws based on nonlinear interpolation with a means of adding or removing grid points accordingly as characteristics either spread in a rarefaction fan or merge into a shock wave. All these methods use the full discretized PDE form and a fixed time step in time integration.

In this paper, we apply the method of lines (MOL) to discretize the PDEs. The advantages and disadvantages of MOL have been discussed by many authors. The moving grid method we chose is the class of methods based on the equidistributing principle (see [3, 5, 10, 11, 22, 23] for detailed information). During discretization using MOL, the most troublesome term is the convection term  $\partial f(u)/\partial x$ . If the central difference scheme, which is adopted by most moving mesh software, is used to discretize this term the odd-order derivatives predominate in the truncation error. The extra term can cause the solution to have an

oscillatory behavior (Gibbs oscillations) that the proper solution did not possess in the vicinity of the transition. Although the moving mesh method can place enough nodes in the wave front and works very well for a convex flux function and sufficiently smooth initial conditions (see the result of [5 or 11] for Burgers' equation), for nonconvex flux functions or piecewise initial conditions, there are still some oscillations appearing in the solution (see the result in Section 4). High-order upwinding schemes are needed in these cases. Almost all the upwinding schemes can be constructed on a nonuniform mesh. We will examine the ENO scheme, the flux-limited scheme, and the PHM scheme. ENO schemes work very well with the constant time-step integration method but have great trouble when solved by adaptive time integration. The flux-limited scheme can be used for adaptive time integration; however, it does not yield good results for some problems. The local third-order PHM method [16], however, can be solved efficiently using a conventional ODE or DAE solver and can yield very good results.

When Eq. (1) is of hyperbolic form (i.e.,  $\mu = 0$ ), a very small artificial viscosity can be added to improve the computational stability. Numerical results show that this does not affect the accuracy of the solution very much. Under the moving mesh frame, an additional convective term,  $u_x \dot{x}$ , appears in the equation. Equation (1) becomes

$$\dot{u} = \frac{\partial f(u)}{\partial x} + \mu u_{xx} + h(u) + u_x \dot{x}$$

$$0 = g(u, x, \dot{x}),$$
(2)

where mesh moving is governed by  $g(u, x, \dot{x})$ .

In Section 2, we describe upwinding schemes which can be used on a nonuniform mesh following the construction methods of Shu and Osher [22] and other authors. In Section 3, we discuss moving mesh methods based on equidistribution. In Section 4, numerical experiments are given to illustrate the advantages of combining the moving mesh with an upwinding scheme and in particular the advantages of the PHM method.

# 2. UPWINDING SCHEMES ON NONUNIFORM MESH

In this section, we consider the discretization of the convective terms on a nonuniform mesh. Unless specified, we consider the general convective term  $f(u)_x$  only. For simplicity, we consider the 1D scalar problem.

Generally, the convective term is discretized into conservation form, e.g.,

$$\frac{\partial f(u)(x))}{\partial x}\bigg|_{x_j} = \frac{2(\hat{f}_{j+1/2} - \hat{f}_{j-1/2})}{x_{j+1} - x_{j-1}},$$
(3)

where  $\hat{f}_{j+1/2}$ , j = 1, 2, ..., n - 1, is called the numerical flux. It is a function of 2k variables

$$\hat{f}_{j+1/2} = \hat{f}(u_{j-k+1}^n, ..., u_{j+k}^n),$$

which satisfies the consistency condition  $\hat{f}(u, ..., u) = f(u)$ . The most commonly used scheme in moving mesh systems, central difference, is also of the form (3), in which  $\hat{f}_{j+1/2} =$  $\frac{1}{2}(f_{j+1}+f_j)$ . For solutions with a rapid transition, the central difference is insufficient to resolve the solution in the transition area. Spurious oscillations often occur in those cases. Many useful schemes have been proposed to solve this problem. Among them, several classes of nonlinear higher order upwinding schemes have been shown to be stable and to avoid spurious oscillations. ENO is one of the most commonly used classes of upwinding schemes to gain high resolution and accuracy with a uniform mesh. It can be of high order and some of them can be used for nonuniform mesh without any modification. The most efficient implementation of ENO methods has been investigated by Shu and Osher in [22]. To be concise, we omit the detailed discussion of Shu and Osher on how to construct high order ENO schemes and give only the explicit form of the second- and third-order ENO schemes following their algorithm ENO-Roe, because to our knowledge, second order or third order is state of the art in moving mesh computation and higher order differencing in moving mesh has proven to be unstable by numerical experiments. In our experience, the first-order Roe scheme [20] can give a good result in many cases and it is the basis for other higher order methods. Thus, we first give the Roe scheme.

## First-Order Roe Scheme

The numerical flux of the Roe scheme is given by

$$\hat{f}_{j+1/2}^{(1)} = \frac{1}{2} (f_j + f_{j+1} - |a_{j+1/2}| (u_{j+1} - u_j)), \tag{4}$$

where  $a_{i+1/2}$  is called the "Roe" speed, which is defined by

$$a_{j+1/2} = \begin{cases} \frac{f_{j+1} - f_j}{u_{j+1} - u_j}, & u_{j+1} \neq u_j \\ \frac{\partial f}{\partial u} \Big|_{u_{j+1/2}}, & u_j = u_{j+1}. \end{cases}$$
(5)

It is easy to see that for the Roe scheme  $\hat{f}_{j+1/2}^{(1)} = f_{k_1}$ , where  $k_1$  is defined by

$$k_1 = \begin{cases} j, & a_{j+1/2} \ge 0\\ j+1, & a_{j+1/2} < 0, \end{cases}$$
(6)

which is in agreement with the principle of upwinding. In

the moving mesh system, there is another convection term  $\dot{x} (\partial u / \partial x)$ . When we want to improve the stability of systems solved by moving mesh methods, this term should also be discretized via upwinding. We give here the first order upwinding scheme,

$$\dot{x} \cdot u_{x} = \begin{cases} \dot{x}_{i} \frac{u_{i+1} - u_{i}}{x_{i+1} - x_{i}}, & \dot{x}_{i} \ge 0, \\ \\ \dot{x}_{i} \frac{u_{i} - u_{i-1}}{x_{i} - x_{i-1}}, & \dot{x}_{i} < 0. \end{cases}$$

$$(7)$$

The upwinding discretization (7) has been shown to improve the stability of the moving mesh system [15]. However, for some problems it does not yield a very accurate solution. We note that when stability is not a problem, the central difference is to be recommended over first-order upwinding because it is more conservative and accurate in those cases.

# Second-Order ENO Scheme

The second-order ENO scheme is based on the firstorder Roe scheme. By choosing the smaller of the firstorder divided differences of f at  $x_{k_1}$ , i.e.,

$$k_{2} = \begin{cases} k_{1} - 1, & \text{if } \left| \frac{f_{k_{1}+1} - f_{k_{1}}}{x_{k_{1}+1} - x_{k_{1}}} \right| \ge \left| \frac{f_{k_{1}} - f_{k_{1}-1}}{x_{k_{1}} - x_{k_{1}-1}} \right| \\ k_{1}, & \text{otherwise.} \end{cases}$$

the estimate of the flux  $\hat{f}_{j+1/2}$  is then

$$\hat{f}_{j+1/2}^{(2)} = f_{k_1} + \left(\frac{f_{k_2+1} - f_{k_2}}{x_{k_2+1} - x_{k_2}}\right) \cdot \frac{1}{2} \left(2x_{j+1/2} - x_{k_1+1/2} - x_{k_1-1/2}\right).$$
(8)

Equation (8) is derived from interpolation at  $x_{k_1}$  (see Shu and Osher [22]). It is second order on a quasi-uniform mesh (see [23]). In practice, we find when replacing the interval  $\frac{1}{2}(2x_{j+1/2} - x_{k_1+1/2} - x_{k_1-1/2})$  with

$$0.5(x_{j+1} - x_j), \quad \text{if } k_1 = j, \tag{9.1}$$

or

$$0.5(x_i - x_{i+1}), \text{ if } k_1 = j+1;$$
 (9.2)

the method is still second order and can be solved more efficiently when combined with the moving mesh.

# Third Order ENO Scheme

The idea of the third order ENO scheme is to compute the third order approximation to f at  $x_{j+1/2}$  in a non-oscillatory way by choosing the smaller of the second order divided differences at  $x_{k_2}$ , i.e., to choose

$$k_3 = \begin{cases} k_2 - 1, & \text{if } |f[x_{k_2+2}, x_{k_2+1}, x_{k_2}]| \ge |f[x_{k_2+1}, x_{k_2}, x_{k_2-1}]| \\ k_2, & \text{otherwise.} \end{cases}$$

Then the numerical flux is

$$f_{j+1/2}^{(3)} = f_{j+1/2}^{(2)} + c[(x_{j+1/2} - x_{k_2+1/2})(x_{j+1/2} - x_{k_2+3/2}) + (x_{j+1/2} - x_{k_2-1/2})(x_{j-1/2} - x_{k_2+3/2}) + (x_{j-1/2} - x_{k_2-1/2})(x_{j+1/2} - x_{k_2+1/2})],$$
(10)

where

$$c = \frac{1}{3} \frac{f[x_{k_3+2}, x_{k_3+1}] - f[x_{k_3+1}, x_{k_3}]}{x_{k_3+2} - x_{k_3}}.$$

Equation (10) is also derived from an interpolation formula and is given by Shu and Osher [22]. From Eq. (3) and Eq. (10), it is easily seen that third-order ENO schemes are functions of four or five variables. Similarly, we can also localize Eq. (10) and restrict the interval to  $[x_{j-1}, x_{j+2}]$ . Equations (8) and (10) appear complicated, however *j* is between  $[k_i - 1, k_i + 1]$  and some terms can be eliminated. Therefore the computational cost is relatively small.

#### Other High Order Upwinding Schemes

There are many other kinds of upwinding schemes. Among them, the flux-limited scheme due to Van Leer [14] and local third-order PHM (piecewise hyperbolic method) scheme due to Marquina [16] can be used in the moving mesh system with slight modification. The second-order flux-limited scheme has been used in the latest NAG library software for first-order partial differential equations (see [18]). The method constructs numerical flux as

$$f_{j+1/2} = \hat{f}_{j+1/2}(u_L, u_R) = \frac{1}{2}(f(u_L) + f(u_R)).$$
(11)

Denote  $h_i = x_i - x_{i-1}$ . Then we have

$$u_L = u_j + \frac{h_{i+1}}{2} \frac{\partial u}{\partial x} \bigg|_{x=x_j} = u_j + \frac{h_{i+1}}{2} \left( \frac{u_j - u_{j-1}}{h_j} \right) \cdot B(r_i),$$

where  $B(r_i)$  is a flux limiter defined by

$$B(r_i) = \frac{r_i + |r_i|}{1 + |r_i|}$$

and

$$r_{i} = \frac{u_{i+1/2}^{c} - u_{i}}{u_{i+1/2}^{u} - u_{i}} = \frac{(u_{i+1} - u_{i})/h_{i+1}}{(u_{i} - u_{i-1})/h_{i}}$$

where  $u_{i+1}^c = \frac{1}{2}(u_i + u_{i+})$  is the central difference term, and  $u_{i+1}^u = u_i + (h_{i+1}/2) ((u_{i+1} - u_i)/h_i)$  is the left upwinding term. Similarly,

$$u_R = u_{i+1} - \frac{h_{i+1}}{2} \frac{(u_{i+2} - u_{i+1})}{h_{i+2}} B\left(\frac{1}{r_{i+1}}\right).$$

The construction of a local PHM scheme was introduced by Marquina [16] for scalar conservation laws. The idea of the PHM scheme is to construct the primitive function of flux via a piecewise hyperbolic equation. Because during our implementation of the moving mesh, we put as many nodes as possible in the transition area, we can simplify the algorithm PHM-REF of Marquina and consider only the nontransition cells. Define  $d_j = f(u(x_j))$  and  $d_{j+1/2} = (d_{j+1} - d_j)/(x_{j+1} - x_j)$  for all *j*. We then assign the central derivative  $d_j$  to be the harmonic mean of  $d_{j-1/2}$  and  $d_{j+1/2}$ 

$$d_j = \frac{2d_{j+1/2}d_{j-1/2}}{d_{j-1/2} + d_{j+1/2}},$$

and  $\alpha$  to be

$$\alpha = \begin{cases} 2\left(\sqrt{\frac{d_j}{d_{j-1/2}}} - 1\right), & \text{if } |d_{j-1/2}| \le |d_{j+1/2}| \\ 2\left(1 - \sqrt{\frac{d_j}{d_{j+1/2}}}\right), & \text{otherwise.} \end{cases}$$

Then the numerical flux can be taken to be

$$\hat{f}_{j+1/2} = \begin{cases} f_j + d_j \cdot h_j \cdot \eta(\alpha), & \text{if } a_{j+1/2} \ge 0, \\ f_{j+1} - d_j \cdot h_j \cdot \eta(-\alpha), & \text{if } a_{j+1/2} < 0, \end{cases}$$
(12)

where  $\eta(x)$  is a differentiable function chosen to ensure that the scheme is third order for a quasi-uniform mesh. This scheme is of second order for a nonuniform mesh. For transition cells  $[x_j, x_{j+1}]$ , we use the Roe scheme. Although technically the Roe scheme is only first order, the moving mesh ensures that the transition cells are very small. In practice, we have observed no loss of global accuracy.

Remark. We give here only the general formula. In

practice, some special cases like  $u_{j+1} = u_j$  can be easily settled separately.

## 3. THE SELF-ADAPTIVE MOVING MESH STRATEGY

As discussed by Furzeland et al. [5], in the classical Lagrange method for fluid-flow problems, the movement of the nodes is attached to a physically motivated, specific flow quantity. For example, for a problem like (1), it makes sense to attach the movement of the nodes to the convection term  $\partial f(u)/\partial x$ , i.e., to choose  $\dot{x} = -df(u)/du$  so as to obtain a parabolic equation of diffusion form. The rationale behind this choice is that parabolic problems without large first-order terms usually possess smoother solutions and thus may be easier to solve numerically. However, the numerical realization of the prescription  $\dot{x} = -df(u)/du$ involves its own difficulties, such as mesh point crossings. In order to obtain a robust moving mesh method which could solve a wide variety of problems, we follow many authors in adopting the equidistribution principle mesh moving strategy with arclength monitor function. We give a brief discussion on how to regularize the mesh moving equation to get a robust and efficient mesh movement and physical solution.

The idea of the equidistribution principle (EP) is to equidistribute a given mesh function in space, placing more nodes where the spatial error is large so as to gain high accuracy overall with a small number of nodes. It is natural to think to incorporate some error measurement into the monitor function. However, in the numerical computation of the moving mesh, high order derivatives cause much numerical noise and instability. Numerical experiments show that at least for finite difference methods, the mesh generated from the error measurement monitor is not very good (see Blom and Verwer [2]).

Regularization is very important for the mesh moving equation (see [15] for detailed information). It has been shown by numerical experiment that without regularization, the Newton iteration for implicit methods often fails to converge and the mesh appears very skew. For explicit methods, without regularization, the computation often exits due to too small time step. The regularization can also be seen as smoothing and diffusing (see [15]), which can generate smoother grids and more accurate solutions.

Given a monitor function M(u, x), the equidistribution principle is

$$\int_{x_i}^{x_i+1} M(x, u) \, dx = \int_{x_i-1}^{x_i} M(x, u) \, dx = \text{Const.}$$

Discretization using the centered integral leads to

$$M(u, x)_{x_{i+1/2}}(x_{x+1} - x_i) = M(x, u)_{x_{i-1/2}}(x_i - x_{i-1}).$$
 (13)

We use the arclength monitor

$$M(u,x)|_{x_{i+1/2}} = \sqrt{1 + \left(\frac{u_{i+1} - u_i}{x_{i+1} - x_i}\right)^2}.$$

Also, the trapezoidal rule can be used in the discretization of the integral. After regularization, we obtain the moving mesh equation

$$\tau A \dot{x} = M x + b, \tag{14}$$

where  $\tau$  is a temporal regularization parameter, A can be viewed as a smoothing operator, and Mx + b is the result of Eq. (13). A can be taken to be several forms besides the identity, each one corresponding to a discretized MMPDE (see [11]). When A is of the form

$$A = \begin{pmatrix} -2 & 1 & & \\ 1 & -2 & & \\ & \ddots & 1 \\ & & 1 & -2 \end{pmatrix},$$

Equation (14) is called MMPDE6 (see [11]). Numerical results in [11] show that the Newton iterations converge faster for MMPDE6 than for other MMPDEs. We will use it in our numerical experiments.

We now discuss the spatial smoothing technique. First, it should be noted that Eq. (13) is not a strict constraint which should be satisfied during computation of the physical solution. It is only a strategy which pushes the nodes to the locations where we want them to be. The discretization of M(x, u) does not affect the accuracy of the physical solution in an analytical sense. But it does affect the mesh distribution. From numerical experiments we know that the mesh quality affects directly the errors due to the spatial discretization, convergence of the Newton iteration for implicit methods, and time step size for explicit methods. Based on Eq. (13), there are two ways in which smoothing can be applied. One is to smooth the mesh x itself, the other is to smooth the monitor. In practice, we often smooth the monitor function. There are two smoothing approaches. One is global smoothing, i.e., smoothing the monitor function over all mesh nodes.

$$\hat{M}_{i} = \sum_{j=1}^{N} \left( \frac{k}{k+1} \right)^{|i-j|} M_{j},$$
(15)

where  $M_i = M(u, x)_{x_{i+1/2}}$ . This approach can restrict the adjacent intervals of mesh nodes to satisfy

$$\frac{k}{k+1} < \max\left(\frac{\Delta x_i}{\Delta x_{i+1}}, \frac{\Delta x_{i-1}}{\Delta x_i}\right) < \frac{k+1}{k}$$

for k > 0. Equation (13) can be turned into

$$\frac{M_i}{n_i}=\frac{M_{i-1}}{n_{i-1}},$$

where  $n_i$  is node concentration defined by

$$n_i = (\Delta x_i)^{-1}, \quad \Delta x_i = x_{i+1} - x_i.$$

Thus, the smoothing (15) can be implemented on node concentration by

$$\hat{n}_i = n_i - k(k+1)(n_{i+1} - 2n_i + n_{i-1}), \quad 1 \le i \le N.$$

This method, called the Dorfi and Drury method, was first proposed in [3], analyzed by Verwer *et al.* [24], and reconsidered by Huang and Russell [12]. In our experience, Eq. (15) results in a quasi-uniform mesh and can yield good results and efficiency if enough nodes are available (see [24] for the needed number of nodes); otherwise, the computation can fail. The other strategy is a local smoothing method, which is implemented by Huang *et al.* [11]. This method restricts the summation of Eq. (15) to its adjacent nodes; i.e.,

$$\hat{M}_{i} = \sum_{j=i-p}^{i+p} \left(\frac{k}{k+1}\right)^{|i-j|} M_{j},$$
(16)

where p is often chosen to be 1 or 2. This strategy does not restrict the mesh too much and also smooths the nodes in the transition corners. However, it often puts too many nodes in the wave fronts or flat areas and is sensitive to the time scale  $\tau$ . Generally, Eq. (16) is used when the number of nodes is small, and the Dorfi and Drury method is used when there are more nodes. Similarly we can impose the smoothing methods on the mesh nodes themselves, which yields

$$\Delta \hat{x}_i = \sum_{j=i-p}^{i+p} \left( \frac{k}{k+1} \right)^{|i-j|} \Delta x_j.$$

This also can have good results.

#### 4. NUMERICAL EXPERIMENTS

Here we consider the comparison of moving mesh methods with central difference and moving mesh methods with upwinding schemes. Throughout, we shall use the arclength monitor. For moving mesh equations of DAE form (e.g.,



FIG. 1. Result of the Dorfi and Drury method for Burgers' equation with 20 nodes and (a) central difference, (b) 2nd-order ENO scheme (9).

Eq. (14) or the Dorfi and Drury method), we solve the systems using the double precision version of the stiff DAE solver DASSL [19] with RTOL = ATOL =  $10^{-6}$ , in which the time integration methods are the backward differentiation formula (BDF), where the approximate Jacobian is computed by DASSL internally using finite differences. Unless specified otherwise, we select the stiffness parameter k = 2 and  $\tau = 0.001$  for the Dorfi and Drury method; for other methods, we choose  $\tau = 0.01$ . All the computations are done on a Sun SPARC-5.

EXAMPLE I. The first example we consider is the famous Burgers' equation

$$\partial u/\partial t = -\partial f(u)/\partial x + \varepsilon \partial^2 u/\partial x^2, \quad 0 < x < 1, t > 0,$$

$$f(u) = u^2/2, \quad \varepsilon = 10^{-4}$$

with piecewise continuous initial condition

$$u(x,0) = \begin{cases} 0.2, & x \le 0.1, \\ 8x - 0.6, & 0.1 \le x \le 0.2, \\ 1.0, & 0.2 \le x \le 0.5, \\ -10x + 6, & 0.5 \le x \le 0.6, \\ 0.0, & 0.6 \le x \le 1.0, \end{cases}$$

and homogeneous Dirichlet conditions at x = 0 and x = 1. This problem has served as a test example for moving mesh methods by Verwer *et al.* [24]. The solution is a wave

where

FIG. 2. Result of the Dorfi and Drury method for Burgers' equation with 40 nodes and central difference.

that steepens and moves to the right until a layer is formed at the end point x = 1. This takes place for  $t \approx 0.2$ . Then the solution slowly decays to zero, near x = 1, because of

1 0.8 0.6 0.4 0.2 ٥ -0.2 0.4 0.6 0.8 0.2

FIG. 3. Result of the Dorfi and Drury method for Burgers' equation with 40 nodes and PHM (12).

TABLE I

Comparison of Different Schemes for Example 1

Scheme	NSTP	NJAC	NRES	CPU
CD	327	77	650	6.43
PHM (12)	368	109	761	10.38
Eq. (9)	1409	2931	1653	82.85
Eq. (11)	556	132	1181	14.87

We have also tested this problem using other schemes and find scheme (8) is prohibitively slow (it takes 4573 time steps and 6361 Jacobian evaluations to reach t = 0.6.); scheme (9) works well before the wave front reaches the left side but takes a much longer time to adjust itself when the shock hits the wall (it takes 203 time steps and 45 Jacobian evaluations before t = 0.6 but takes 1386 time steps and 1650 Jacobian evaluations to reach t = 1.0.); scheme (11) works faster than schemes (8) and (9) but yields a poor solution (even worse than central difference). The solution results for schemes (8) and (9) are very similar to PHM scheme (12). The computational efficiency (with 40 nodes) can be seen from Table I, where CD denotes the central difference, NSTP denotes the number of time steps used, NRES denotes the number of calls to the residual subroutine and NJAC denotes the number of Jacobian evaluations.

EXAMPLE II. For convection-diffusion problems with convex flux (convection term), the moving mesh with central difference works well when the number of nodes is great enough or sufficient smoothing is used. To test the moving mesh methods on more difficult problems, we consider the Riemann problem with nonconvex flux,

$$u_t + (f(u))_x = \varepsilon \partial u^2 / \partial^2 x, \quad -1 < x < 1, t > 0, \varepsilon = 10^{-4}, \quad (23)$$

where

$$f(u) = \frac{1}{4}(u^2 - 1)(u^2 - 4).$$

The initial condition is given by

$$u(x, 0) = \begin{cases} -3, & \text{if } x \le 0, \\ 3, & \text{if } x > 0. \end{cases}$$

This problem has been chosen as a test example for ENO schemes by many authors [22], [16]. For initial condition (23), instead of the expected sonic rarefaction fan that appears for convex fluxes, a nonconvex sonic stationary shock at x = 0 develops. Our choice of scheme solves this problem well. In Figs. (4)–(7), we compare the numerical resolution of (23) solved with central difference, 2nd-order







FIG. 4. Result of moving mesh with central difference for Problem II.

ENO scheme (9), 2nd-order flux limited scheme (11) and PHM scheme (12). We solved this problem using the MMPDE6 ([11]) and DASSL. We observe that for the central difference, there is a spurious oscillation. However, the results for the upwinding schemes are good. The PHM scheme is the best. The advantages of the PHM scheme can also be seen from Table II (with 40 nodes).

I ABLE II						
Comparison	of Different Sc	chemes for	Example 2			

Scheme	NSTP	NJAC	NRES	CPU
CD	7148	1224	12678	144.283
PHM	216	31	409	5.7833



FIG. 5. Result of moving mesh with 2nd-order ENO scheme (9) for Problem II.



FIG. 6. Result of moving mesh with flux limiter scheme (11) for Problem II.

*Remark.* The moving mesh schemes we have considered here all result in a DAE, with stiffness depending on the original PDE, the mesh spacing, and the time scale  $\tau$ . We have solved this DAE using implicit numerical methods. For nonstiff systems, explicit methods can be more efficient. Moving mesh methods which result in an ODE which can be solved efficiently by explicit

methods have been developed in [13]. Upwinding schemes can also be used advantageously in combination with these methods. We performed the same numerical experiments using the moving mesh method of [13] and the explicit ODE solver DDRIVE (down load from netlib) by Kahaner and Sutherland, with quite similar results.



FIG. 7. Result of moving mesh with local PHM scheme (12) for Problem II.

## 5. CONCLUSION

Several upwinding schemes have been described for the convection term of the moving mesh. Some strategies and techniques are also discussed. They are applied to different types of problems. The numerical results are very encouraging and clearly show that in many cases the moving mesh with upwinding scheme can track the shock or the wave front more accurately than the commonly used central difference. Of course, a wide range of experiments should be done. However, from the examples we have given, we can see that combining the moving mesh with an upwinding scheme can make those methods which failed because of spurious oscillations work well now and can track the rarefaction as well as the shock for problems with nonconvex flux.

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